

Recall:

First Law:

$$\boxed{dU = \delta Q_{\text{in}} + \delta W_{\text{in}}}$$

Allowing only pV work interaction

$$\delta W_{\text{in}} = -pdV$$

Second Law:

$$\boxed{dS = \left(\frac{\delta Q}{T}\right) + \delta \mathcal{P}_s \quad \text{where} \quad \delta \mathcal{P}_s \geq 0}$$

If internally reversible (i.e., production equals zero)

$$dS = \left(\frac{\delta Q}{T}\right)_{\text{IntRev}} \quad \text{or} \quad TdS = \delta Q_{\text{IntRev}}$$

Combining

$$dU = TdS - pdV$$

rearranging:

$$TdS = dU - pdV \tag{1}$$

or

$$dS = \frac{1}{T}dU + \frac{p}{T}dV$$

and on a per unit mass basis:

$$\boxed{ds = \frac{1}{T}du + \frac{p}{T}dv} \tag{2}$$

Similarly, with the definition of enthalpy:

$$H \equiv U + pV$$

and differentiating (using the product rule):

$$dH = dU + pdV + Vdp$$

Combining with Eq. (1) yields:

$$TdS = dH - Vdp \tag{3}$$

or

$$dS = \frac{1}{T}dH - \frac{V}{T}dp$$

and on a per unit mass basis:

$$\boxed{ds = \frac{1}{T}dh - \frac{v}{T}dp} \tag{4}$$

Both Eqs. (2) and (4) are general (reversible or irreversible, assuming only pV work and non-reacting).

In general,  $h = h(T, p)$ , so:

$$dh = \left.\frac{\partial h}{\partial T}\right|_p dT + \left.\frac{\partial h}{\partial p}\right|_T dp$$

For a non-reaction gas, if we further assume that the gas is thermally perfect (i.e.,  $u = u(T)$  and  $h = h(T)$ ) and recalling that  $c_v \equiv \left. \frac{\partial u}{\partial T} \right|_v$  and  $c_p \equiv \left. \frac{\partial h}{\partial T} \right|_p$ , we may write:

$$du = c_v dT \quad \text{and} \quad dh = c_p dT$$

Note that  $c_v = c_v(T)$  and  $c_p = c_p(T)$ .

For a thermally perfect gas:

$$pv = RT \quad \text{or} \quad pV = mRT \quad \text{or} \quad pV = nKT \quad \text{or} \quad pv = \frac{R_u}{M} T$$

In other words:

$$\frac{p}{T} = \frac{R}{v} \quad \text{and} \quad \frac{v}{T} = \frac{R}{p}$$

Equation (2) may be then written:

$$\boxed{ds = \frac{1}{T} c_v dT + \frac{R}{v} dv} \quad (5)$$

and Eq. (4) may be written:

$$\boxed{ds = \frac{1}{T} c_p dT - \frac{R}{p} dp} \quad (6)$$

If we know  $c_v(T)$  and  $c_p(T)$ , Eqs. (5) and (6) may be integrated from some reference state (see for example Table A-22 for integrated properties of air).

$$\Delta s = s_2 - s_1 = \int_1^2 \frac{1}{T} c_p(T) dT - R \ln \frac{p_2}{p_1} = s^0(T_2) - s^0(T_1) - R \ln \frac{p_2}{p_1}$$

**TABLE A-22 Ideal Gas Properties of Air**

| <i>T(K), h and u (kJ/kg), s° (kJ/kg · K)</i> |          |          |           |                       |                      |          |          |          |           |                      |                      |
|--|----------|----------|-----------|-----------------------|----------------------|----------|----------|----------|-----------|----------------------|----------------------|
| <i>T</i>                                     | <i>h</i> | <i>u</i> | <i>s°</i> | when $\Delta s = 0^1$ |                      | <i>T</i> | <i>h</i> | <i>u</i> | <i>s°</i> | when $\Delta s = 0$  |                      |
|  |          |          |           | <i>p<sub>r</sub></i>  | <i>v<sub>r</sub></i> |          |          |          |           | <i>p<sub>r</sub></i> | <i>v<sub>r</sub></i> |
| 200  | 199.97   | 142.56   | 1.29559   | 0.3363                | 1707.                | 450      | 451.80   | 322.62   | 2.11161   | 5.775                | 223.6                |
| 210  | 209.97   | 149.69   | 1.34444   | 0.3987                | 1512.                | 460      | 462.02   | 329.97   | 2.13407   | 6.245                | 211.4                |
| 220  | 219.97   | 156.82   | 1.39105   | 0.4690                | 1346.                | 470      | 472.24   | 337.32   | 2.15604   | 6.742                | 200.1                |
| 230  | 230.02   | 164.00   | 1.43557   | 0.5477                | 1205.                | 480      | 482.49   | 344.70   | 2.17760   | 7.268                | 189.5                |
| 240  | 240.02   | 171.13   | 1.47824   | 0.6355                | 1084.                | 490      | 492.74   | 352.08   | 2.19876   | 7.824                | 179.7                |
| 250  | 250.05   | 178.28   | 1.51917   | 0.7329                | 979.                 | 500      | 503.02   | 359.49   | 2.21952   | 8.411                | 170.6                |
| 260  | 260.09   | 185.45   | 1.55848   | 0.8405                | 887.8                | 510      | 513.32   | 366.92   | 2.23993   | 9.031                | 162.1                |
| 270  | 270.11   | 192.60   | 1.59634   | 0.9590                | 808.0                | 520      | 523.63   | 374.36   | 2.25997   | 9.684                | 154.1                |
| 280  | 280.13   | 199.75   | 1.63279   | 1.0889                | 738.0                | 530      | 533.98   | 381.84   | 2.27967   | 10.37                | 146.7                |
| 285  | 285.14   | 203.33   | 1.65055   | 1.1584                | 706.1                | 540      | 544.35   | 389.34   | 2.29906   | 11.10                | 139.7                |
| 290  | 290.16   | 206.91   | 1.66802   | 1.2311                | 676.1                | 550      | 554.74   | 396.86   | 2.31809   | 11.86                | 133.1                |
| 295  | 295.17   | 210.49   | 1.68515   | 1.3068                | 647.9                | 560      | 565.17   | 404.42   | 2.33685   | 12.66                | 127.0                |
| 300  | 300.19   | 214.07   | 1.70203   | 1.3860                | 621.2                | 570      | 575.59   | 411.97   | 2.35531   | 13.50                | 121.2                |
| 305  | 305.22   | 217.67   | 1.71865   | 1.4686                | 596.0                | 580      | 586.04   | 419.55   | 2.37348   | 14.38                | 115.7                |
| 310  | 310.24   | 221.25   | 1.73498   | 1.5546                | 572.3                | 590      | 596.52   | 427.15   | 2.39140   | 15.31                | 110.6                |
| 315  | 315.27   | 224.85   | 1.75106   | 1.6442                | 549.8                | 600      | 607.02   | 434.78   | 2.40902   | 16.28                | 105.8                |
| 320  | 320.29   | 228.42   | 1.76690   | 1.7375                | 528.6                | 610      | 617.53   | 442.42   | 2.42644   | 17.30                | 101.2                |
| 325  | 325.31   | 232.02   | 1.78249   | 1.8345                | 508.4                | 620      | 628.07   | 450.09   | 2.44356   | 18.36                | 96.92                |
| 330  | 330.34   | 235.61   | 1.79783   | 1.9352                | 489.4                | 630      | 638.63   | 457.78   | 2.46048   | 19.84                | 92.84                |
| 340  | 340.42   | 242.82   | 1.82790   | 2.149                 | 454.1                | 640      | 649.22   | 465.50   | 2.47716   | 20.64                | 88.99                |
| 350  | 350.49   | 250.02   | 1.85708   | 2.379                 | 422.2                | 650      | 659.84   | 473.25   | 2.49364   | 21.86                | 85.34                |
| 360  | 360.58   | 257.24   | 1.88543   | 2.626                 | 393.4                | 660      | 670.47   | 481.01   | 2.50985   | 23.13                | 81.89                |
| 370  | 370.67   | 264.46   | 1.91313   | 2.892                 | 367.2                | 670      | 681.14   | 488.81   | 2.52589   | 24.46                | 78.61                |
| 380  | 380.77   | 271.69   | 1.94001   | 3.176                 | 343.4                | 680      | 691.82   | 496.62   | 2.54175   | 25.85                | 75.50                |
| 390  | 390.88   | 278.93   | 1.96633   | 3.481                 | 321.5                | 690      | 702.52   | 504.45   | 2.55731   | 27.29                | 72.56                |
| 400  | 400.98   | 286.16   | 1.99194   | 3.806                 | 301.6                | 700      | 713.27   | 512.33   | 2.57277   | 28.80                | 69.76                |
| 410  | 411.12   | 293.43   | 2.01699   | 4.153                 | 283.3                | 710      | 724.04   | 520.23   | 2.58810   | 30.38                | 67.07                |
| 420  | 421.26   | 300.69   | 2.04142   | 4.522                 | 266.6                | 720      | 734.82   | 528.14   | 2.60319   | 32.02                | 64.53                |
| 430  | 431.43   | 307.99   | 2.06533   | 4.915                 | 251.1                | 730      | 745.62   | 536.07   | 2.61803   | 33.72                | 62.13                |
| 440  | 441.61   | 315.30   | 2.08870   | 5.332                 | 236.8                | 740      | 756.44   | 544.02   | 2.63280   | 35.50                | 59.82                |

1. *p<sub>r</sub>* and *v<sub>r</sub>* data for use with Eqs. 6.41 and 6.42, respectively.

Figure 1: Thermally perfect gas properties of air

Example: What is the enthalpy of air (per unit mass) at 660 K and 100 atm?

Table A-22  $\Rightarrow h = 670.47 \text{ kJ/kg}$

CoolProp at 1 Pa  $\Rightarrow h = 671.044 \text{ kJ/kg}$

CoolProp at 1 atm  $\Rightarrow h = 671.051 \text{ kJ/kg}$

CoolProp at 100 atm  $\Rightarrow h = 672.334 \text{ kJ/kg}$

Percent change from 1 atm to 100 atm = 0.1908%, not very much.

Example: What is the entropy of air (per unit mass) at 660 K and 100 atm?

Table A-22  $\Rightarrow s^0 = 2.50985 \text{ kJ}/(\text{kg K})$

But wait, that is not  $s$ !

$$s(T, p) = s^0(T) - R \ln \frac{p}{p_{\text{ref}}}$$

Recall  $R_{\text{air}} = 0.287 \text{ kJ/kg}$  and, often,  $p_{\text{ref}}$  is taken as 1 atm.

So, for our example:

$$s(T, p) = 2.50985 - 0.287 \ln \frac{100}{1} = 1.18817 \frac{\text{kJ}}{\text{kg} \cdot \text{K}}$$

Typically, though, we are interested in the change of entropy during a process:

Example: What is the change in entropy of air (per unit mass) from 300 K and 0.7 atm to 660 K and 100 atm?

Using:

$$s_2(T_2, p_2) - s_1(T_1, p_1) = s_2^0(T_2) - s_1^0(T_1) - R \ln \frac{p_2}{p_1} \quad (7)$$

Yielding:

$$s_2(T_2, p_2) - s_1(T_1, p_1) = 2.50985 - 1.70203 - 0.287 \ln \frac{100}{0.7} = -0.61623 \frac{\text{kJ}}{\text{kg} \cdot \text{K}}$$

CoolProp  $\Rightarrow \Delta s = -0.625063 \text{ kJ}/(\text{kg K})$

Isentropic process (TPG)

If a process is idealized as adiabatic and reversible, then it is isentropic. For thermally perfect gases these processes can be simplified from Eq. (7), setting the left-hand side to zero. Dividing by  $R$ , exponentiating, and rearranging:

$$\frac{p_2}{p_1} = \frac{\exp [s^0(T_2)/R]}{\exp [s^0(T_1)/R]} = \frac{p_{r2}}{p_{r1}}$$

where  $p_{r1} \equiv \exp [s^0(T_1)/R]$  is the relative pressure (NOT the reduced pressure used in our compressibility diagrams).

Using the thermally perfect gas state equation we may also write:

$$\frac{v_2}{v_1} = \left[ \frac{RT_2}{p_{r2}} \right] \left[ \frac{p_{r1}}{RT_1} \right] = \frac{v_{r2}}{v_{r1}}$$

Entropy change for a calorically perfect gas:

If we further assume that  $c_v(T)$  and  $c_p(T)$  are nearly constant (i.e., a calorically perfect gas), then the integrations is easy:

$$s_2 - s_1 = c_v \ln \left( \frac{T_2}{T_1} \right) + R \ln \left( \frac{v_2}{v_1} \right) \quad (8)$$

and

$$s_2 - s_1 = c_p \ln \left( \frac{T_2}{T_1} \right) - R \ln \left( \frac{p_2}{p_1} \right) \quad (9)$$

Often, we study idealized processes which are internally reversible and adiabatic, and therefore, isentropic.

Isentropic relationships (isentropic, TPG, and CPG)

For processes that are internally reversible and adiabatic and if the working fluid may be modeled as both thermally and calorically perfect, then we may derive some very simple but extremely useful relationships.

Consistent with our current assumptions, we may start from Eq. (8) or Eq. (9). For an isentropic process, Eq. (8) may be written:

$$0 = c_v \ln \left( \frac{T_2}{T_1} \right) + R \ln \left( \frac{v_2}{v_1} \right)$$

or

$$\frac{c_v}{R} \ln \left( \frac{T_2}{T_1} \right) = \ln \left( \frac{v_1}{v_2} \right) = \ln \left( \frac{\rho_2}{\rho_1} \right) \quad (10)$$

Given the ratio of specific heats,  $\gamma \equiv c_p/c_v$  and that for a thermally perfect gas:

$$c_p = c_v + R$$

we find that:

$$\frac{c_v}{R} = \frac{1}{\gamma - 1}$$

As a result, Eq. (10) may be written:

$$\frac{1}{\gamma - 1} \ln \left( \frac{T_2}{T_1} \right) = \ln \left( \frac{v_1}{v_2} \right) = \ln \left( \frac{\rho_2}{\rho_1} \right)$$

Exponentiating:

$$\left( \frac{T_2}{T_1} \right)^{\left( \frac{1}{\gamma-1} \right)} = \left( \frac{v_1}{v_2} \right) = \left( \frac{\rho_2}{\rho_1} \right) \quad (11)$$

Similarly, an isentropic process and Eq. (9) yields:

$$0 = c_p \ln \left( \frac{T_2}{T_1} \right) - R \ln \left( \frac{p_2}{p_1} \right)$$

or

$$\frac{c_p}{R} \ln \left( \frac{T_2}{T_1} \right) = \ln \left( \frac{p_2}{p_1} \right)$$

With the definition of  $\gamma$  and the relation  $c_p = c_v + R$ , we find:

$$\frac{c_p}{R} = \frac{\gamma}{\gamma - 1}$$

Substituting and exponentiating yields:

$$\boxed{\left(\frac{T_2}{T_1}\right)^{\left(\frac{\gamma}{\gamma-1}\right)} = \left(\frac{p_2}{p_1}\right)} \quad (12)$$

Combining Eqs. (11) and (13), we can find a third isentropic relationships:

$$\boxed{\left(\frac{p_2}{p_1}\right) = \left(\frac{\rho_2}{\rho_1}\right)^\gamma} \quad (13)$$

Example:

Air initially at 300 K and 1 atm is compressed in a reversible, adiabatic process to 1/10 its initial volume.

Find the final temperature and pressure of the air.